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Adiabatic calorimetric study on shape memory alloys: heat capacities and martensitic phase transformation of Ni₄₇Ti₄₄Nb₉ and Nb₈₀Ti₁₄Ni₆ alloys^{$\dot{\alpha}$}

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Abstract

A low-temperature precision adiabatic calorimeter for heat capacity measurements has been constructed. The heat capacities of a shape memory alloy $(Ni_{47}Ti_{44}Nb_{9})$ and a normal reference alloy ($Nb_{80}Ti_{14}Ni_6$) have been measured in the temperature range from 60 to 350 K with this calorimeter. The heat capacity curve of the normal alloy is smooth and continuous over the whole temperature range. An anomaly, indicating the martensitic phase transformation, in the heat capacity curve of the shape memory alloy was observed at $T = 234.0 \pm 0.5$ K with a maximum, where $C_{p,m} = 34.86 \pm 0.21 \text{ J K}^{-1} \text{ mol}^{-1}$. The molar enthalpy and molar entropy associated with the martensitic transformation were determined as $\Delta_{\text{trs}} H_{\text{m}}^{\circ} = 411.9 \pm 2.5 \text{ J} \text{ mol}^{-1}$ and $\Delta_{\text{trs}} S_{\text{m}}^{\circ} = 1.76 \pm 0.01 \text{ J K}^{-1} \text{ mol}^{-1}$, respectively.

Keywords: Adiabatic calorimetry; Heat capacity; Martensitic phase transformation; Shape memory alloy

1. Introduction

The shape memory effect (SME) has received increasing attention in recent decades, because shape memory alloys (SMA) have been widely used as new functional materials

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with commercial applications in pipe couplings, electrical connectors, various actuators, medical appliance, etc. The shape memory effect is one of the effects associated with martensitic phase transformations $\lceil 1-3 \rceil$ induced by stress or thermal treatment. These effects include thermoelasticity, pseudoelasticity, SME and the two-way shape memory effect. After a material is deformed by application of a stress at one temperature, the shape memory effect arises when it recovers its original shape during heating to a higher temperature instead of removing the stress. Many alloy systems such as CuZnAl [4], TiPd [S], FePt [6] and TiNi [7-lo] show SME, and TiNi alloy is the most widely used SMA because of its excellent mechanical properties among these systems.

Martensitic transformation proceeds in equilibrium between the chemical and nonchemical driving forces. These two kinds of forces can be evaluated separately in a quantitative way using heat capacity data [ll], so that more information concerning the martensitic phase transformation can be obtained. Although differential scanning calorimetry has been used to measure the heat capacities of SMA [121 as well as to determine the transformation temperatures $[5, 13, 14]$, no research work on SMA by adiabatic calorimetry has been reported until now. For this purpose we have constructed an adiabatic calorimeter, and measured the heat capacities of a shape memory alloy and a reference alloy in the temperature range from 60 to 350 K using this calorimeter. On the basis of the heat capacity measurements, the molar enthalpy and molar entropy associated with the martensitic transformation of the investigated SMA were determined.

2. **Experimental**

2.1. *Materials and preparation of alloy samples*

Samples of the two alloys studied were prepared by the conventional vacuum induction melting technique. Titanium (purity, 99.8%) nickel (purity, 99.9%) and niobium (purity, 99.9%) were remelted in a graphite crucible and cast in vacuum. The ingots were homogenized at 800°C for 24 h in an argon atomosphere and then quenched in water. Then, they were machined to a cylinder 20 mm in diameter for the heat capacity measurements. The atomic compositions of the alloys prepared were $Ni_{47}Ti_{44}Nb_{9}$ for the shape memory alloy (SMA) sample and $Nb_{80}Ti_{14}Ni_{6}$ for the reference alloy sample. Measured by a four-wire electrical resistance method prior to the heat capacity measurements, the transformation temperatures of the SMA sample were found to be as follows: $M_s = 170 \text{ K}$, $M_f = 148 \text{ K}$, $A_s = 225 \text{ K}$ and $A_f = 251 \text{ K}$, where *M,* and *M,* are the start and finish temperatures of a forward martensitic transformation, and A_s and A_f are the start and finish temperatures of a reverse martensitic transformation.

2.2. Construction of the adiabatic calorimeter

The heat capacity measurements of the two alloys were performed by a lowtemperature adiabatic calorimeter. The construction of the cryostat of the calorimeter is similar to that described elsewhere [151. The alloy sample, sealed in a sample cell and

surrounded by two adiabatic shields, was housed in a high vacuum and immersed in liquid nitrogen. The two adiabatic shields were made of chromium-plated copper (0.5 mm thick) and equipped with manganin heating wires (0.2 mm in diameter). The inner shield, which is a little smaller than the outer one, was hung from the top of the outer one by three pieces of fine nylon thread, while the outer shield was hung from the top of the vacuum can by the same means. Two sets of six-junction chromel-cope1 (Ni 55%; Cu, 45%) thermocouples were used to indicate the temperature differences between the sample cell and the inner adiabatic shield, and between the inner and the outer shields. The temperature of the two shields were controlled separately with two sets of autoadiabatic controllers operating in PID mode. The principle of the automatic adiabatic control circuitry was described in detail previously $[15, 16]$. The vacuum can was evacuated to about 1×10^{-3} Pa by an oil diffusion pump system during heat capacity measurements. A mechanical pump was used to pump out the nitrogen vapour from the Dewar vessel to solidify the remaining liquid nitrogen, and then 60 K or an even lower temperature was obtained.

The sample cell, shown in Fig. 1, was made of gold-plated copper (0.3 mm wall thickness) with an internal volume of 6.3 cm^3 . On the side surface of the cell, a heater of Karma wires (0.15 mm in diameter, $R \approx 105\Omega$) was wound bifilarly and fixed with

Fig. 1. Sample cell of the calorimeter: A, copper capillary; B, gold-plated silver lid; C, sealing flange; D, main body of the sample cell; E, aluminium-plated dacron film; F, heating wires.

a special cycloweld, and covered with a piece of aluminium-plated dacron film. A copper sheath was silver-soldered on the bottom of the cell to contain a miniature platinum resistance thermometer. The thermometer (Model IPRT, No. 2, SIAM Co., 15 mm long, 1.5 mm in diameter, $R_0 = 100 \Omega$) was calibrated in terms of ITS-90 at the Centre of Low-Temperature Metrology and Measurements, Academia Sinica, in the temperature range 50-400 K with an uncertainty of about 1 mK. A gold-plated copper lid with a length of copper capillary was sealed to the sample cell using a small amount of the cycloweld. After the alloy sample was loaded in the cell, 0.1 MPa helium gas was introduced to the cell through the capillary to facilitate thermal equilibrium, and then the capillary was pinched off. The sample cell was hung from the top of the inner adiabatic shield by a length of nylon thread through a hook made by the capillary. The apparatus for temperature and energy measurement used in this calorimeter are the same as used in our previous work $[16]$.

2.3. *Experimental procedure*

The alloy sample, sealed in the sample cell, was first cooled down to the lowest temperature (about 55 K which is the lower limit of the temperature that the calorimetric system can reach) and kept at this temperature for half an hour. The heat capacity measurements as a function of temperature in the heating direction using the standard discrete heating method were then initiated. In this intermittent mode, the sequence of heat capacity measurements were composed of two successively alternating periods. One was "the drift period" during which the temperature of the calorimeter cell was observed as a function of time. The other was "the energy input period" during which a definite amount of electrical energy was supplied to the calorimeter cell. The supplied energy was determined on the basis of measurements of voltage and current across and through the heater of the sample cell as well as the interval of energy input. The cycles of measurements for temperature and energy input were repeated in the above sequence until the upper limit of the temperature (about 350 K) for the measurements had been reached. In our experiments the duration of energy input was 10 min, and the thermal equilibrium inside the sample cell was attained within 3–5 min after the energy input. The temperature increment for each experimental point was about 3 K in the whole temperature range from 60 to 350 K.

3. **Results and discussion**

3.1. Heat capacities of α -Al₂O₃

In order to assess the precision and the accuracy of the present calorimeter, heat capacity measurements on the standard reference material α -Al₂O₃ were made before the measurements on the alloy samples. The α -Al₂O₃ sample was a spectroscopically pure reagent with a purity of 99.993 wt%. XRD analysis showed that the whole sample was in the α -phase. The mass of the α -Al₂O₃ sample used for the measurements amounted to 8.7353 g (0.085673 mol, based on a molar mass of 101.9613 g mol⁻¹). The

Fig. 2. Plot of deviations $100 \times (\Delta C_{p,m}/C_{p,m,ref})$ of the experimental values for the molar heat capacity of α -Al₂O₃, where $C_{p,m,ref}$ is the molar heat capacity regarded as the reference value given by Ditmars et al. [17] and $\Delta C_{p,m}$ is the difference between our value and $C_{p,m,re}$

| T/K | $C_{p,m}$ | T/K | $C_{p,m}/$ | T/K | $C_{p,m}$ | T/K | $C_{p,m}$ | |
|---------|------------------------------|---------|------------------------------|---------|------------------------------|---------|------------------------------|--|
| | $J K^{-1}$ mol ⁻¹ | |
| 62.664 | 8.399 | 133.144 | 18.507 | 209.594 | 25.252 | 275.881 | 26.746 | |
| 65.776 | 9.047 | 136.426 | 18.832 | 213.198 | 26.301 | 279.718 | 26.614 | |
| 68.863 | 9.666 | 139.849 | 19.071 | 216.609 | 27.468 | 283.578 | 26.570 | |
| 71.951 | 10.262 | 143.414 | 19.369 | 219.854 | 29.029 | 287.447 | 26.516 | |
| 75.158 | 10.870 | 146.926 | 19.605 | 222.928 | 30.557 | 291.291 | 26.475 | |
| 78.498 | 11.484 | 150.397 | 19.836 | 225.836 | 32.416 | 295.107 | 26.431 | |
| 81.764 | 12.078 | 153.824 | 20.098 | 228.633 | 33.581 | 298.922 | 26.399 | |
| 85.131 | 12.651 | 157.347 | 20.313 | 231.323 | 34.450 | 302.739 | 26.363 | |
| 88.603 | 13.247 | 160.968 | 20.557 | 233.999 | 34.854 | 306.543 | 26.367 | |
| 92.050 | 13.779 | 164.549 | 20.789 | 236.642 | 34.509 | 310.345 | 26.371 | |
| 95.601 | 14.317 | 168.162 | 21.014 | 239.333 | 33.624 | 314.144 | 26.378 | |
| 99.213 | 14.838 | 171.811 | 21.237 | 242.084 | 32.794 | 317.944 | 26.386 | |
| 102.780 | 15.313 | 175.500 | 21.440 | 244.905 | 31.784 | 321.750 | 26.395 | |
| 106.377 | 15.736 | 179.228 | 21.673 | 247.822 | 30.945 | 325.550 | 26.408 | |
| 110.010 | 16.162 | 182.997 | 21.928 | 250.845 | 30.091 | 329.339 | 26.422 | |
| 113.539 | 16.594 | 186.805 | 22.130 | 253.553 | 29.025 | 333.109 | 26.437 | |
| 116.961 | 16.977 | 190.566 | 22.379 | 257.120 | 28.270 | 336.867 | 26.453 | |
| 120.296 | 17.277 | 194.367 | 22.729 | 260.485 | 28.050 | 340.614 | 26.469 | |
| 123.567 | 17.619 | 198.192 | 23.114 | 264.024 | 27.690 | 344.347 | 26.485 | |
| 126.775 | 17.970 | 202.020 | 23.625 | 267.769 | 27.441 | 348.073 | 26.501 | |
| 129.926 | 18.208 | 205.849 | 24.273 | 272.067 | 27.024 | 351.813 | 26.519 | |

Table 1 Experimental molar heat capacity of shape memory alloy; $M(Ni_{47}Ti_{44}Nb_{9}) = 57.0160$ g mol⁻¹

measured $C_{p,m}$ values were fitted to a sixth-degree polynomial. The overall precision of this calorimeter was \pm 0.2%, as evaluated from the deviations of the experimental $C_{p,m}$ values from the sixth-degree polynomial. Fig. 2 shows the deviations of the measured molar heat capacities of α -Al,O₁ from the recommended values measured by Ditmars et al. [17]. Expect for some points near 110 K, our results agreed with the literature values within \pm 0.6%, as shown in the plot.

3.2. *Heat capacities of SMA and RA*

The experimental molar heat capacities of the shape memory alloy $(Ni_{47}Ti_{44}Nb_{9})$ and the reference alloy $(Nb_{80}Ti_1_4Ni_6)$ samples are listed in Tables 1 and 2, respectively. The masses of the samples used in heat capacity measurements were 38.9340 g (0.68286 mol) for SMA and 23.9619 g (0.28340 mol) for RA. The molar masses of the substances investigated in this paper were calculated based on the Standard Atomic Weights of Elements 1991 [18]. The temperature range of measurements covered $60-350$ K for both alloy samples. The heat capacity of the samples represented 79% of the measured total heat capacity at 60 K to 82% at 350 K for SMA, and 67 to 64% for RA. The temperature increment due to each energy input was small enough to permit the curvature correction to be ignored in comparison with the experimental error.

| T/K | | | | | | | $C_{\mathfrak{p},\mathfrak{m}}/$ $J K^{-1}$ mol ⁻¹ |
|---------|---|---------|--|---------|---|---------|--|
| | $C_{p,m}$ $J K^{-1}$ mol ⁻¹ | T/K | $C_{p,m}/$ $J K^{-1}$ mol ⁻¹ | T/K | $C_{p,m}$ / $J K^{-1}$ mol ⁻¹ | T/K | |
| 59 362 | 10.120 | 133.819 | 20.213 | 209.469 | 23.262 | 288.168 | 24.758 |
| 61.260 | 10.605 | 137.370 | 20.421 | 213.168 | 23.340 | 291.916 | 24.816 |
| 63.437 | 11.118 | 140.859 | 20.622 | 216.847 | 23.416 | 295.616 | 24.876 |
| 66.003 | 11.665 | 144.353 | 20.891 | 220.502 | 23.492 | 299.326 | 24.942 |
| 69.238 | 12.313 | 147.909 | 21.045 | 224.131 | 23.568 | 303.053 | 25.002 |
| 73.208 | 13.076 | 151.483 | 21.228 | 227.741 | 23.643 | 306.837 | 25.056 |
| 77.662 | 13.872 | 155.021 | 21.421 | 231.342 | 23.717 | 310.675 | 25.117 |
| 82.134 | 14.658 | 158.580 | 21.567 | 234.928 | 23.791 | 314.508 | 25.180 |
| 86.552 | 15.374 | 162.212 | 21.740 | 238.448 | 23.865 | 318.338 | 25.233 |
| 90.787 | 16.017 | 165.867 | 21.853 | 242.072 | 23.938 | 322.160 | 25.296 |
| 94.701 | 16.582 | 169.495 | 22.015 | 245.821 | 24.011 | 325.976 | 25.357 |
| 98.505 | 17.062 | 173.097 | 22.180 | 249.735 | 24.082 | 329.786 | 25.414 |
| 102.242 | 17.577 | 176.673 | 22.311 | 253.636 | 24.153 | 333.592 | 25.472 |
| 105.955 | 17.976 | 180.279 | 22.414 | 257.526 | 24.224 | 337.397 | 25.539 |
| 109.516 | 18.312 | 183.918 | 22.523 | 261.392 | 24.294 | 341.190 | 25.601 |
| 112.958 | 18.555 | 187.534 | 22.661 | 265.230 | 24.363 | 344.971 | 25.656 |
| 116.380 | 18.867 | 191.181 | 22.779 | 269.075 | 24.432 | 348.742 | 25.720 |
| 119.788 | 19.237 | 194.864 | 22.868 | 272.935 | 24.501 | 352.498 | 25.778 |
| 123.229 | 19.444 | 198.529 | 22.980 | 276.766 | 24.568 | | |
| 126.712 | 19.721 | 202.173 | 23.064 | 280.581 | 24.635 | | |
| 130.242 | 19.973 | 205.802 | 23.166 | 284.381 | 24.701 | | |

Table 2 Experimental molar heat capacity of the reference alloy; $M(Nb_{80}Ti_{14}Ni_6) = 84.5503$ g mol⁻¹

Fig. 3. Measured molar heat capacities of SMA and RA. \bullet , SMA (Ni₄₇Ti₄₄Nb₉); \bigcirc , RA (Nb₈₀Ti₁₄Ni₆).

A plot of the experimental values of $C_{p,m}$ versus T is shown in Fig. 3. From the figure, it can be seen that the curve of $C_{p,m}$ vs. *T* for RA sample is completely smooth, with no sign of any phase transformation or other thermal anomaly in the whole temperature range between 60 and 350 K. The curve of $C_{p,m}$ vs. *T* for SMA, however, clearly shows a thermal anomaly which is associated with the martensitic phase transformation. The heat capacity anomaly corresponding to the martensitic transformation began at approx. 209 K, reached its maximum at (234.0 \pm 0.5) K with a value of C_{p,m} = (34.86) $+$ 0.21) J K⁻¹ mol⁻¹, and ended at approx. 265 K, respectively. Assuming that the normal heat capacity in the transition temperature region was approximated by the smoothed interpolating curve, the molar enthalpy and the molar entropy associated with the martensitic transformation were determined as $\Delta_{\text{trs}}H_{\text{m}}^{\circ} = 411.9 \pm 2.5 \text{ J mol}^{-1}$ and $\Delta_{\text{trs}} S_{\text{m}}^{\circ} = 1.76 \pm 0.01 \text{ J K}^{-1} \text{ mol}^{-1}$, respectively, by graphical integration of the excess heat capacities,

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